



# Drones Demystified!

K. Alexis, C. Papachristos, Autonomous Robots Lab, University of Nevada, Reno

A. Tzes, Autonomous Robots & Intelligent Systems Lab, NYU Abu Dhabi

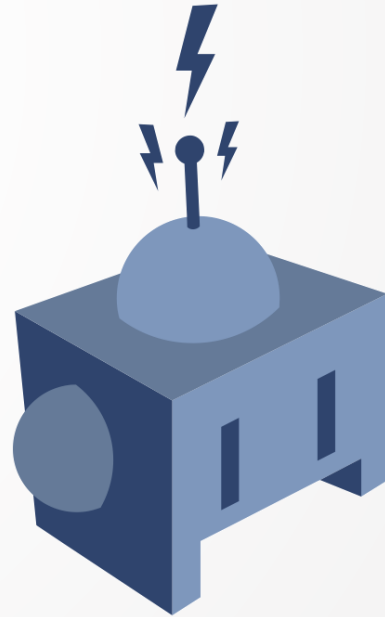
A decorative graphic on the left side of the slide, featuring a blue arrow pointing right and several thin, curved lines in shades of blue and grey.

# Drones Demystified!

## Topic: Propulsion Systems

# Propulsion Systems for Robotics

How do I  
move?



# Propulsion Systems for Robotics

- Different propulsion systems designs are employed for different robotic configurations.
- Miniaturization of propulsion systems –in combination with good efficiency– is among the reasons for the success of small robotics.
- Within this course we will focus on:
  - DC Motors
  - DC Brushless Motors
  - Propelled-systems



# Our focus

## Design of Propelled Aerial Robots

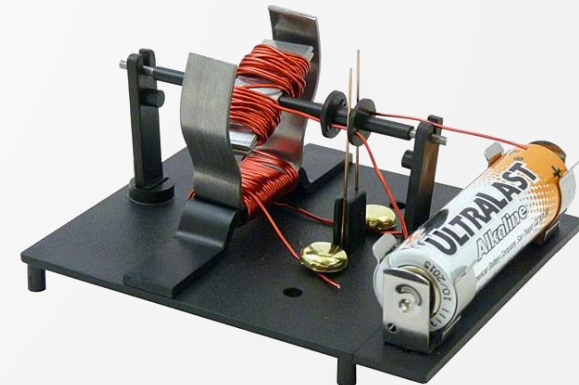
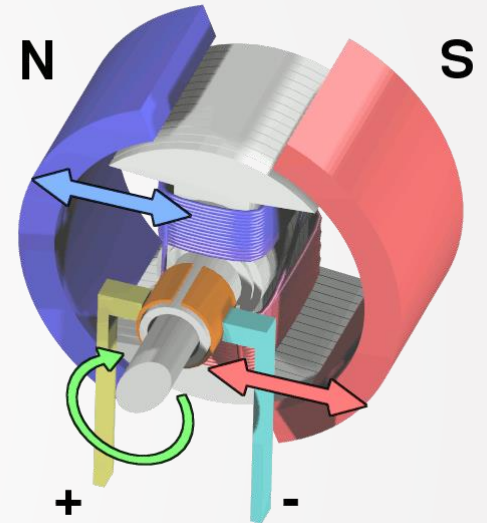
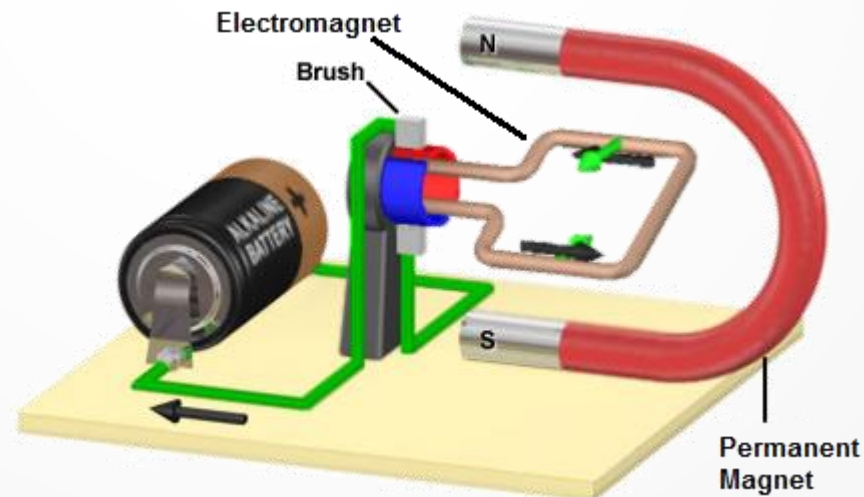


**Many other interesting configurations also exist!**



# DC Motors

- Stationary permanent magnet
- Electromagnet on axis induces torque
- Split ring + brushes switch direction of current
- If you have never built one – do so!



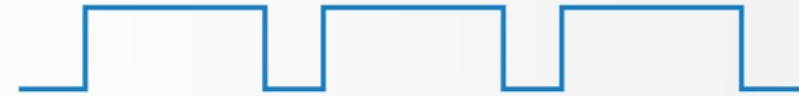
# Control of DC Motors

- More power means faster rotation – how to conveniently control power in a digital sense?
- How to modulate power using a digital signal? What is the digital equivalent function to directly control power at the input?
  - Fixed voltage with pulse modulation – specifically:
- Pulse Width Modulation (PWM)
- Duty cycle is the proportion of “ON” time vs. period

50% duty cycle



75% duty cycle

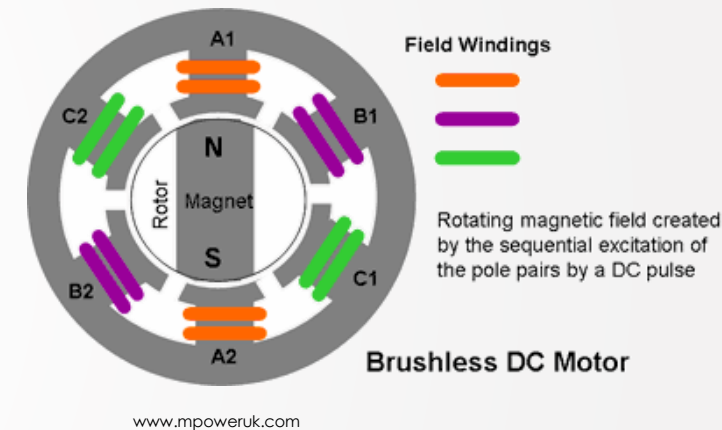


25% duty cycle



# Brushless Motors

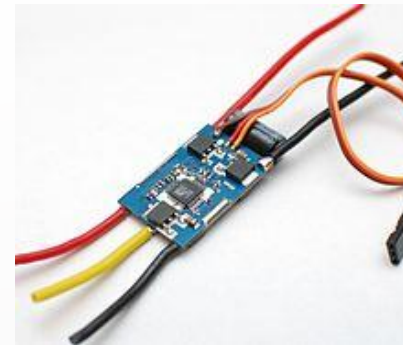
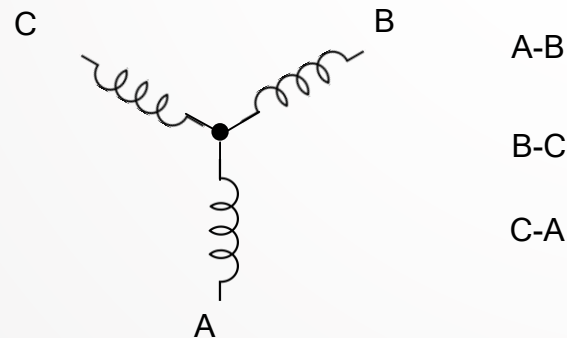
- ▶ Electromagnets are stationary
- ▶ Permanent magnets on the axis (either inside or outside)
- ▶ Three coils (or more)
- ▶ No brushes (less maintenance, higher efficiency)
- ▶ Brushless motors come with high torque, mostly eliminating the need for gearboxes in case of multicopter aerial robots therefore maximizing endurance



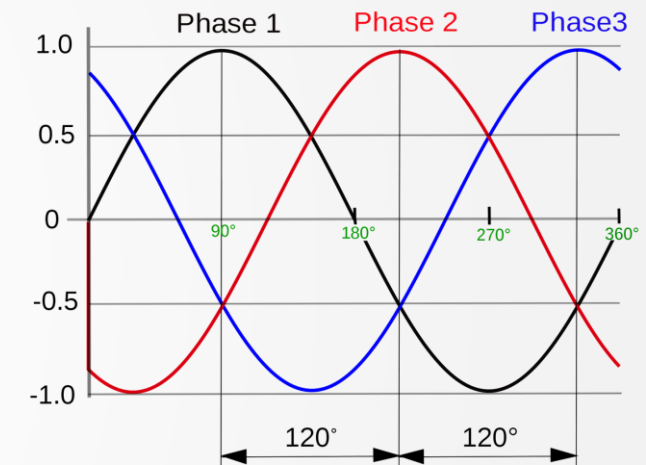


# Brushless Electronic Speed Controllers

- Typically one microcontroller per motor
  - Called Electronic Speed Controller (ESC)
- Generates PWM signal for the three motor phases
- AC signal converter (MOSFET) to convert PWM to analogue output
- Measure motor position/speed using back-EMF

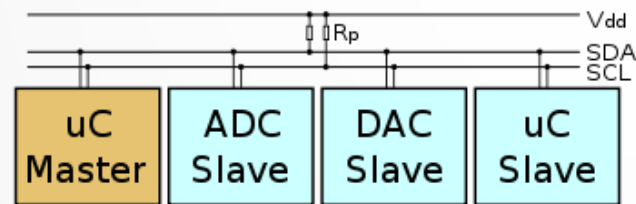


[http://en.wikipedia.org/wiki/File:ESC\\_35A.jpg](http://en.wikipedia.org/wiki/File:ESC_35A.jpg)

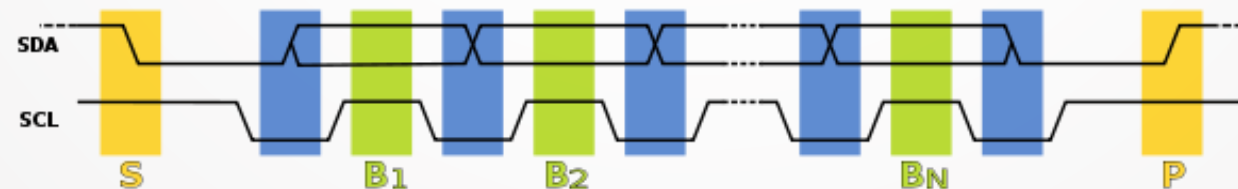


# I2C Protocol

- Often digital protocol to command the ESC
- Serial data line (SDA) + serial clock line (SCL)
  - Specific encoding/decoding allows master/slave communication
- All devices connected in parallel
- 7-10 bit address, 100-3400 kbit/s speed
- Communication between motor controller and autopilot



<http://en.wikipedia.org/wiki/File:I2C.svg>



[http://en.wikipedia.org/wiki/File:I2C\\_data\\_transfer.svg](http://en.wikipedia.org/wiki/File:I2C_data_transfer.svg)

# The Micro Aerial Vehicle propeller



➤ Is something much simpler than a helicopter rotor

# The Micro Aerial Vehicle propeller

**Air-Flow Visualization for  
Several Rotor Configuration  
Models Operating in the Static  
Thrust Condition.**

- ▶ Video of airflow and vortex patterns with propellers. These tests were conducted at NACA, now NASA Langley Research Center. The interior tests were probably at the Propeller Research Tunnel. The exterior tests at the end of the film were at the Helicopter Test Tower. Langley Film #L-118



# The Micro Aerial Vehicle propeller

- Rotor modeling is a very complicated process.
- **A Rotor is different than a propeller.** It is not-rigid and contains degrees of freedom. Among them blade flapping allows the control of the rotor tip path plane and therefore control the helicopter.



- Used to produce thrust.
- Propeller plane perpendicular to shaft.
- Rigid blade. No flapping.
- Fixed blade pitch angle or collective changes only.



- Used to produce lift and directional control.
- Elastic element between blade and shaft.
- Blade flapping used to change tip path plane.
- Blade pitch angle controlled by swashplate.





# The Micro Aerial Vehicle propeller

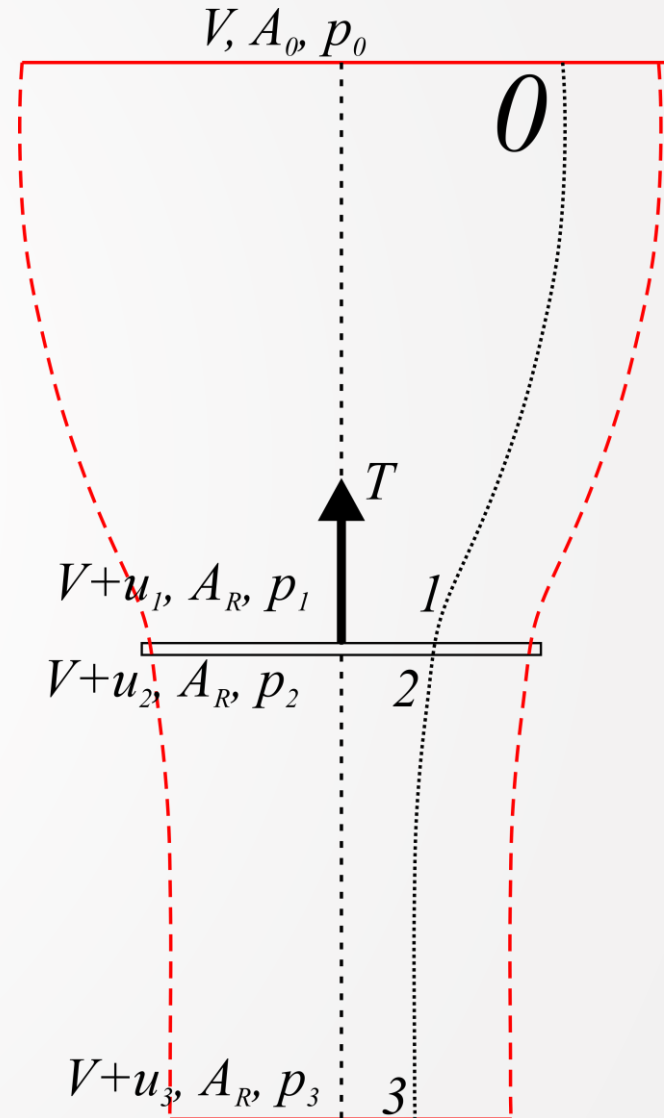
- In a simplified assumption, a propeller is considered to present no blade flapping.
- It is approximated as a rotor disc producing thrust and drag forces.
- Thrust & Power Equations

$$F_{Thrust} = \frac{1}{2} \rho A v^2$$

$$P = \frac{1}{2} A v^2$$

- Hover case (ideal power):

$$P = \frac{F_{Thrust}^{3/2}}{\sqrt{2\rho A_R}} = \frac{(mg)^{3/2}}{\sqrt{2\rho A_R}}$$



# The Micro Aerial Vehicle propeller

## Thrust & Power Equations

$$F_{Thrust} = \frac{1}{2} \rho A v^2$$

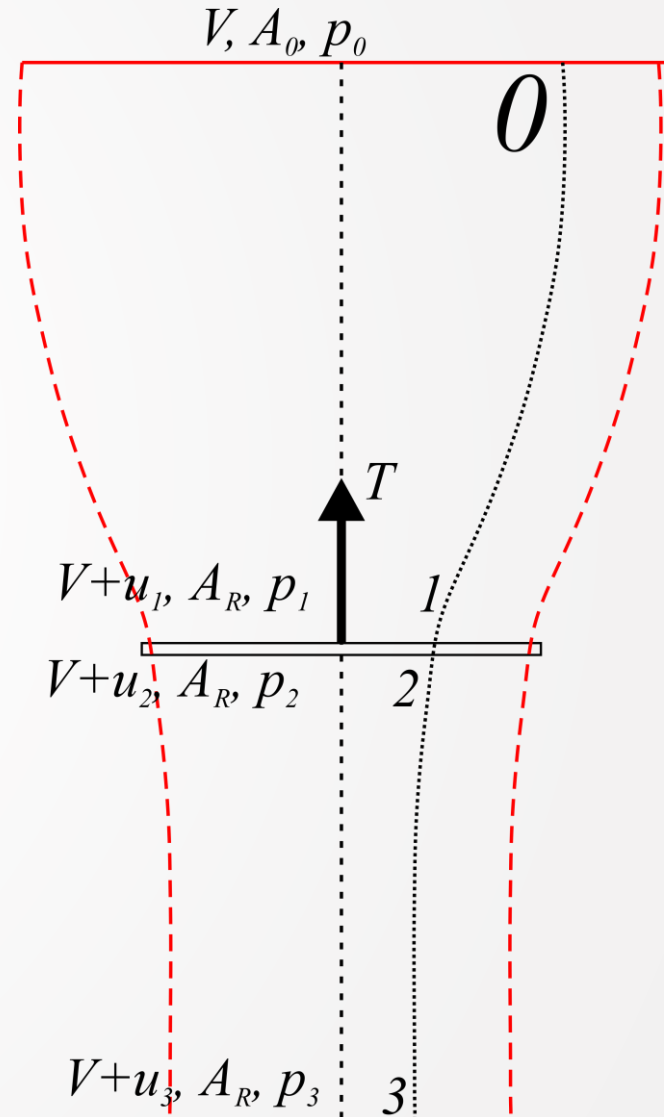
$$P = \frac{1}{2} \rho A v^3$$

## Hover case (ideal power):

$$P = \frac{F_{Thrust}^{3/2}}{\sqrt{2\rho A_R}} = \frac{(mg)^{3/2}}{\sqrt{2\rho A_R}}$$

## Figure of Merit:

$$FM = \frac{\text{Ideal Power to Hover}}{\text{Real Power to Hover}}$$



# The Micro Aerial Vehicle propeller

## ► Lift & Drag at Blade Element:

$$dL = \frac{\rho}{2} C_L c dr V^2 \quad dD = \frac{\rho}{2} C_D c dr V^2$$

$$dT = N_b (dL \cos \phi - dD \sin \phi)$$

$$dQ = N_b (dL \sin \phi + dD \cos \phi) r$$

$$V \approx V_T \quad \phi \approx \frac{V_P}{V_T}$$

$$dT \approx N_b dL$$

$$C_L = C_{L\alpha} (\alpha - \alpha_0)$$

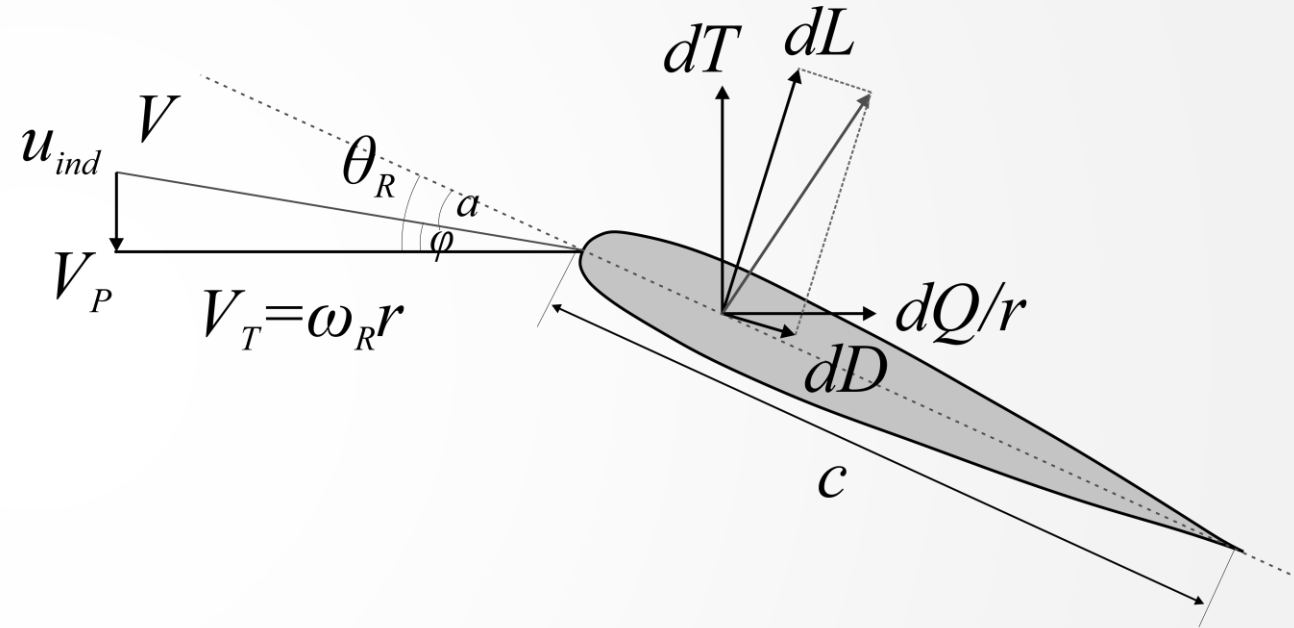
$$C_{L\alpha} = 2\pi$$

$$C_{D\alpha} = 5.7$$

## ► A0: zero lift angle of attack.

## ► Linearize polar for Reynolds number at 2/3 R

$$dT_{be} = N_b \frac{\rho}{2} C_{L\alpha} \left( \theta_R - \frac{V_P}{V_T} - \alpha_0 \right) c dr V_T^2$$



# The Micro Aerial Vehicle propeller

- Simplified model forces and moments:
  - **Thrust Force:** the resultant of the vertical forces acting on all the blade elements.

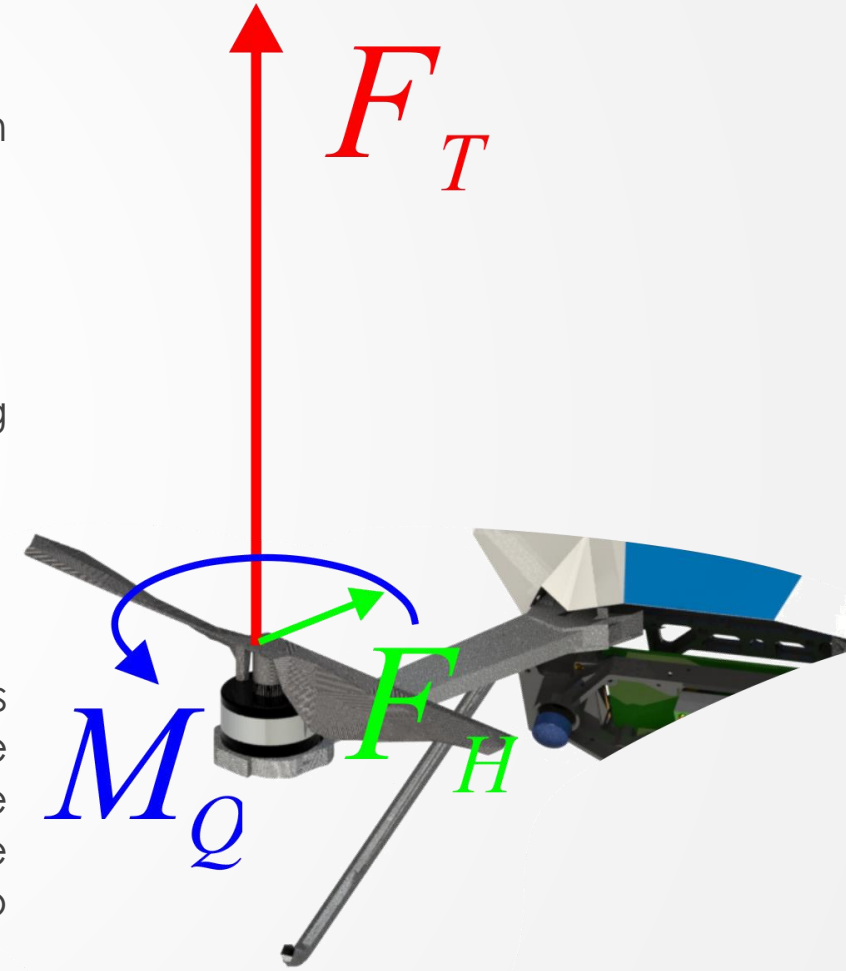
$$F_T = T = C_T \rho A (\Omega R)^2$$

- **Hub Force:** the resultant of all the horizontal forces acting on all the blade elements.

$$F_H = H = C_H \rho A (\Omega R)^2$$

- **Drag Moment:** This moment about the rotor shaft is caused by the aerodynamic forces acting on the blade elements. The horizontal forces acting on the rotor are multiplied by the moment arm and integrated over the rotor. Drag moment determines the power required to spin the rotor.

$$M_Q = Q = C_Q \rho A (\Omega R)^2 R$$



# The wheel of a small ground robot

## ► Circular Motion – Rotational Formulas

### ► Angular Velocity

$$\omega = \theta/t \quad v = \omega r$$

### ► Angular Velocity and Acceleration

$$\omega = \omega_0 + at$$

### ► Angular Displacement

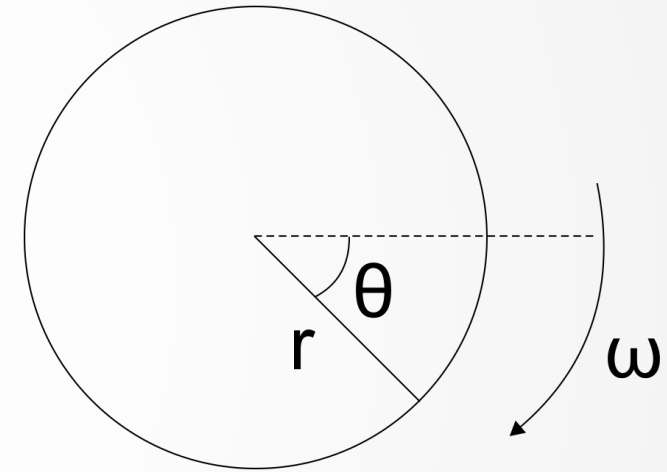
$$\theta = \omega_0 t + \frac{1}{2}at^2$$

### ► Angular Acceleration

$$a = \frac{d^2\theta}{dt^2} = \frac{d\omega}{dt}$$

### ► Angular Momentum or Torque

$$T = aJ_w$$



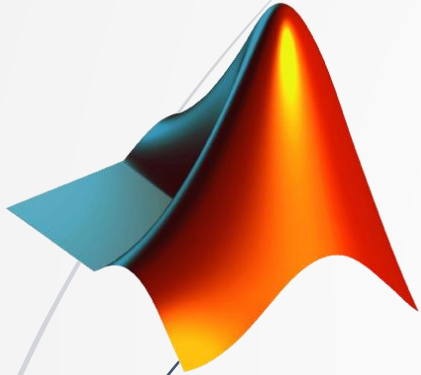
- $\omega$  = angular velocity
- $\theta$  = angular position
- $r$  = radius of the wheel
- $a$  = angular acceleration
- $J_w$  = moment inertia
- $T$  = angular momentum



# Code Example

## ▶ MATLAB DC Motor Control Example

- ▶ [https://github.com/unr-arl/drones\\_demystified/tree/master/matlab/propulsion-systems/motor-control](https://github.com/unr-arl/drones_demystified/tree/master/matlab/propulsion-systems/motor-control)
- ▶ MATLAB 2016 Live note



A black and white photograph of a drone flying in front of a construction site. The drone is in the foreground, slightly out of focus, with its four rotors visible. In the background, several large construction cranes are visible, also out of focus, against a bright sky. The overall scene is a construction site.

**Thank you!**

Please ask your question!