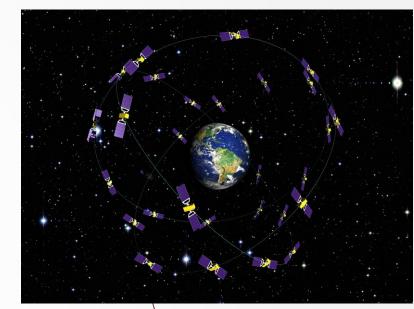






Global Positioning System

- 24 Satellites orbiting the Earth (and some back-ups).
- Altitude set at 20,180km
- Any point on Earth's surface can be seen by at least 4 satellites at all times.
- Time-of-Flight of radio signal from 4 satellites to receiver in 3 dimensions.
- 4 range measurements needed to account for clock offset error.
- 4 nonlinear equations in 4 unknown results:
 - Latitutde
 - Longitutde
 - Altitude
 - Receiver clock time offset





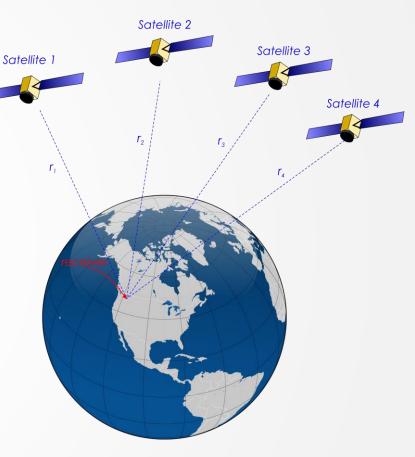
Global Positioning System

Time-of-Flight of the radio signal from satellite to receiver used to calculate pseudorange.

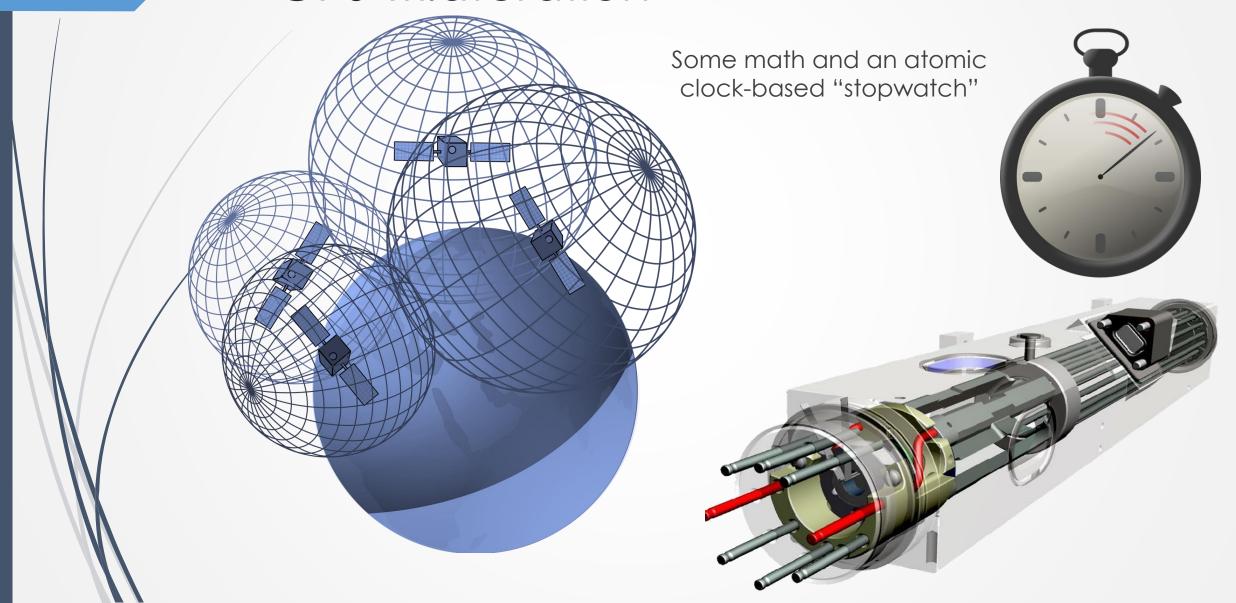
Called pseudorage to distinguish it from true range.

Numerous sources of error in time-of-flight measurement:

- Ephemeris Data errors in satellite location
- Satellite clock due to clock drift.
- Ionosphere upper atmosphere, free electrons slow transmission of the GPS signal.
- Troposphere lower atmosphere, weather (temperature and density) affect speed of light, GPS signal transmission.
- Multipath Reception signals not following direct path
- Receiver measurement limitations in accuracy of the receiver timing.
- Small timing errors can result in large position deviations:
 - 10ns timing error leads to 3m pseudorange error.



GPS Trilateration



GPS Error Characterization

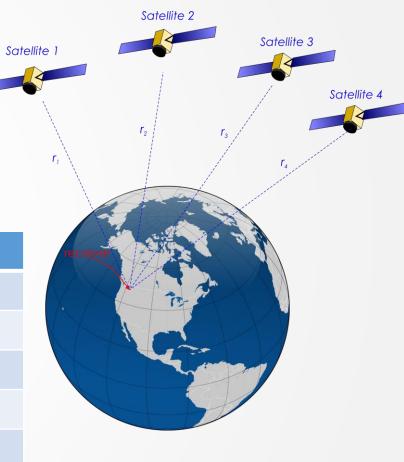
Cumulative effect of GPS pseudorange errors is described by the User-Equivalent Range Error (UERE).

UERE has two components:

- Bias
- Random

 1σ , in m

Error source	Bias	Random	Total
Ephemeris data	2.1	0.0	2.1
Satellite clock	2.0	0.7	2.1
Ionosphere	4.0	0.5	4.0
Troposphere monitoring	0.5	0.5	0.7
Multipath	1.0	1.0	1.4
Receiver measurement	0.5	0.2	0.5
UERE, rms	5.1	1.4	5.3
Filtered UERE, rms	5,1	0.4	5.1



GPS Error Characterization

 Effect of satellite geometry on position calculation is expressed by dilution of precision (DOP).

Satellites close together leads to high DOP.

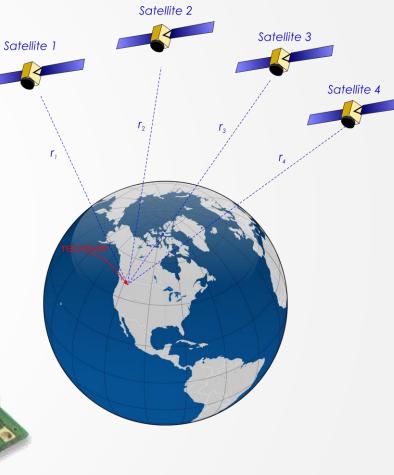
Satellites far apart leads to low DOP.

DOP varies with time.

Horizontal DOP (HDOP) is smaller than Vertical DOP (VDOP):

- Nominal HDOP = 1.3
- Nominal VDOP = 1.8





Total GPS Error

Standard deviation of RMS error in the north-east plane:

$$E_{n-e,rms} = \text{HDOP} \times \text{UERE}_{rms} \Rightarrow$$

 $E_{n-e,rms} = (1.3)(5.1) = 6.6\text{m}$

Standard deviation of RMS altitude error:

$$E_{h,rms} = \text{VDOP} \times \text{UERE}_{rms} \Rightarrow$$

 $E_{h,rms} = (1.8)(5.1) = 9.2\text{m}$

As expected: an ellipsoidal error model.

Further categorization

Let's categorize the sensors we overviewed.

	Absolute		Rate	
	GPS Barometer Accelerometer Magnetometer		Airspeed senso Gyroscope	or
	Position		Orientation	
/	GPS Airspeed Barometer		Accelerometer Gyroscope Magnetometer	
	Sensor	Measures		Predicts
	Accelerometer	Extracts orientation and measures acceleration		Velocity

SenseSoar

Flight Tests:

Dynamics and Onboard Avionics Evaluation

May, 24th 2013

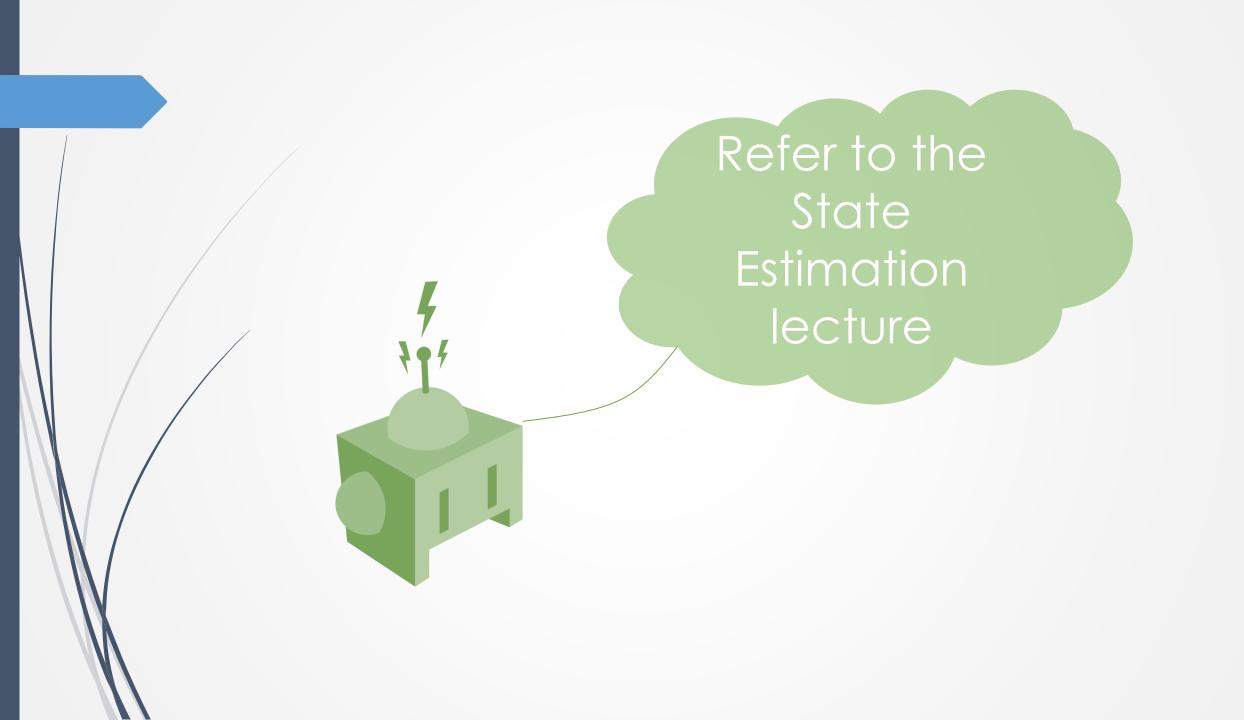


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Find out more

http://www.autonomousrobotslab.com/literature-and-links.html

