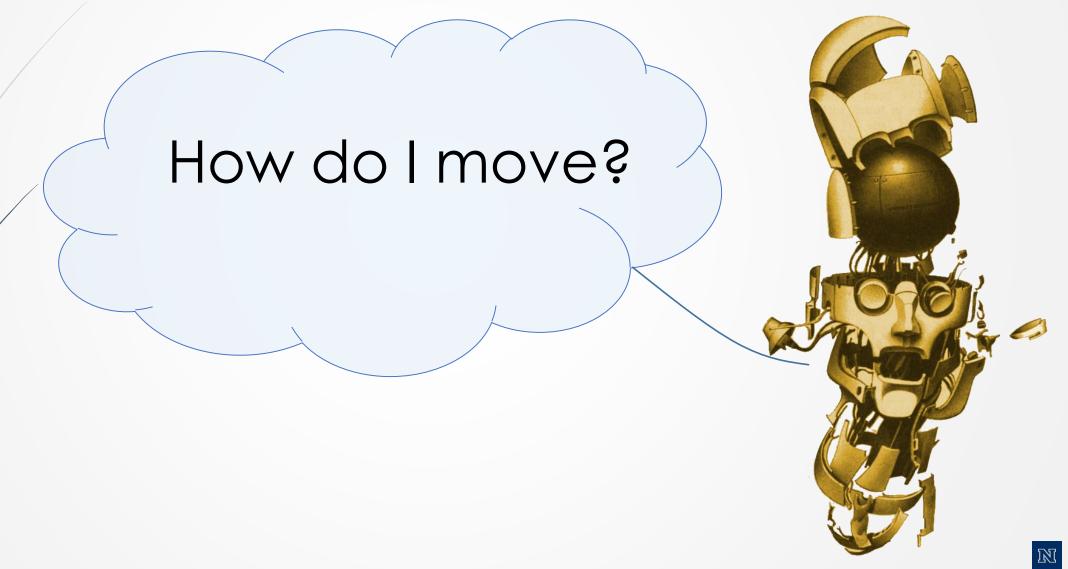


# Autonomous Mobile Robot Design

Topic: Recap Vehicle Modeling, State Estimation

Dr. Kostas Alexis (CSE)

# Autonomous Robot Challenges





# Autonomous Mobile Robot Design Topic: Vehicle Propulsion

Dr. Kostas Alexis (CSE)

# The Micro Aerial Vehicle propeller

- Simplified model forces and moments:
  - Thrust Force: the resultant of the vertical forces acting on all the blade elements.

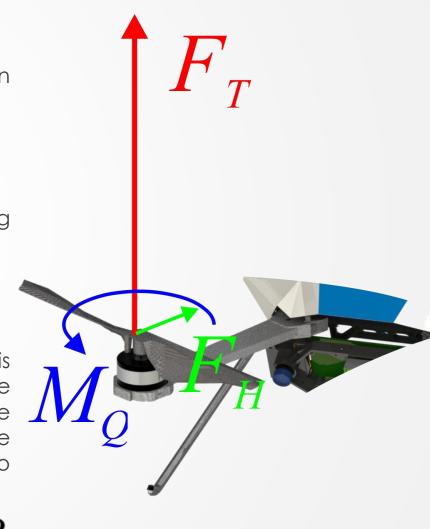
$$F_T = T = C_T \rho A(\Omega R)^2$$

Hub Force: the resultant of all the horizontal forces acting on all the blade elements.

$$F_H = H = C_H \rho A(\Omega R)^2$$

■ Drag Moment: This moment about the rotor shaft is caused by the aerodynamic forces acting on the blade elements. The horizontal forces acting on the rotor are multiplied by the moment arm and integrated over the rotor. Drag moment determines the power required to spin the rotor.

$$M_Q = Q = C_Q \rho A(\Omega R)^2 R$$



# The wheel of a small ground robot

#### Circular Motion – Rotational Formulas

- Angular Velocity  $\omega = \theta/t \quad v = \omega r$
- Angular Velocity and Acceleration  $\omega = \omega_0 + at$
- Angular Displacement
  1
  1
  1
  2

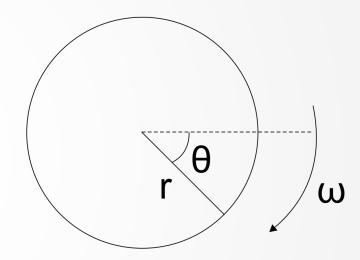
$$\theta = \omega_0 t + \frac{1}{2} a t^2$$

Angular Acceleration

$$a = \frac{d^2\theta}{dt} = \frac{d\omega}{dt}$$

Angular Momentum or Torque

$$T = aJ_w$$



- ω = angular velocity
- $\theta$  = angular position
- r = radius of the wheel
- a = angular acceleration
- $J_w = moment inertia$
- T = angular momentum

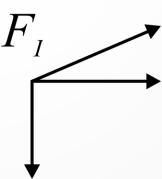


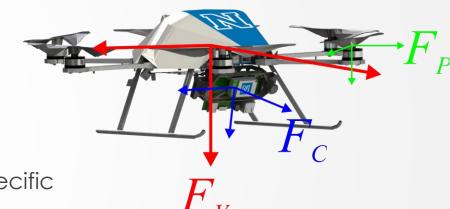
# Autonomous Mobile Robot Design Topic: Coordinate Frame Transformations

Dr. Kostas Alexis (CSE)

### Coordinate Frames

- In Guidance, Navigation and Control of aerial robots, reference coordinate frames are fundamental.
- Describe the relative position and orientation of:
  - Aerial Robot relative to the Inertial Frame
  - On-board Camera relative to the Aerial Robot body
  - Aerial Robot relative to Wind Direction
- Some expressions are easier to formulate in specific frames:
  - Newton's law
  - Aerial Robot Attitude
  - Aerodynamic forces/moments
  - Inertial Sensor data
  - GPS coordinates
  - Camera frames





## Rotation of Reference Frame

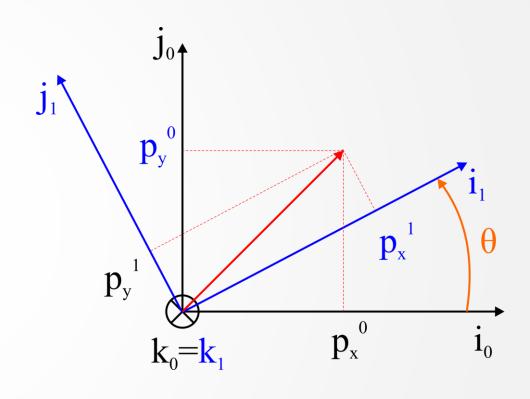
Rotation around the k-axis

$$\mathbf{p} = p_x^0 \mathbf{i}^0 + p_y^0 \mathbf{j}^0 + p_z^0 \mathbf{k}^0$$

$$\mathbf{p} = p_x^1 \mathbf{i}^1 + p_y^1 \mathbf{j}^1 + p_z^1 \mathbf{k}^1$$

$$\mathbf{p}^1 = \begin{pmatrix} p_x^1 \\ p_y^1 \\ p_z^1 \end{pmatrix} = \begin{pmatrix} \mathbf{i}^1 \mathbf{i}^0 & \mathbf{i}^1 \mathbf{j}^0 & \mathbf{i}^1 \mathbf{k}^0 \\ \mathbf{j}^1 \mathbf{i}^0 & \mathbf{j}^1 \mathbf{j}^0 & \mathbf{j}^1 \mathbf{k}^0 \\ \mathbf{k}^1 \mathbf{i}^0 & \mathbf{k}^1 \mathbf{j}^0 & \mathbf{k}^1 \mathbf{k}^0 \end{pmatrix} \begin{pmatrix} p_x^0 \\ p_y^0 \\ p_z^0 \end{pmatrix}$$

$$\mathbf{p}^{1} = \mathcal{R}_{0}^{1} \mathbf{p}^{0}, \ \mathcal{R}_{0}^{1} = \begin{bmatrix} \cos \theta & \sin \theta & 0 \\ -\sin \theta & \cos \theta & 0 \\ 0 & 0 & 1 \end{bmatrix}$$



### Rotation of Reference Frame

Rotation around the i-axis

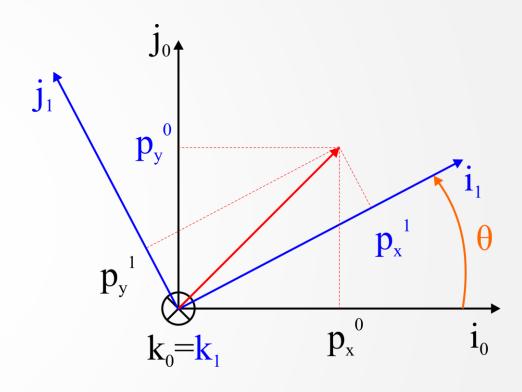
$$\mathcal{R}_0^1 = \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos \theta & \sin \theta \\ 0 & -\sin \theta & \cos \theta \end{bmatrix}$$

Rotation around the j-axis

$$\mathcal{R}_0^1 = \begin{bmatrix} \cos \theta & 0 & -\sin \theta \\ 0 & 1 & 0 \\ \sin \theta & 0 & \cos \theta \end{bmatrix}$$

Rotation around the k-axis

$$\mathcal{R}_0^1 = \begin{bmatrix} \cos \theta & \sin \theta & 0 \\ -\sin \theta & \cos \theta & 0 \\ 0 & 0 & 1 \end{bmatrix} \qquad \begin{array}{c} \bullet & (\mathcal{R}_a^c) & 1 = (\mathcal{R}_a^c) \\ \bullet & \mathcal{R}_b^c \mathcal{R} a^b = \mathcal{R}_a^c \\ \bullet & \det(\mathcal{R}_a^b) = 1 \end{array}$$



Orthonormal matrix properties

$$(\mathcal{R}_a^b)^- 1 = (\mathcal{R}_a^b)^T = \mathcal{R}_b^a$$

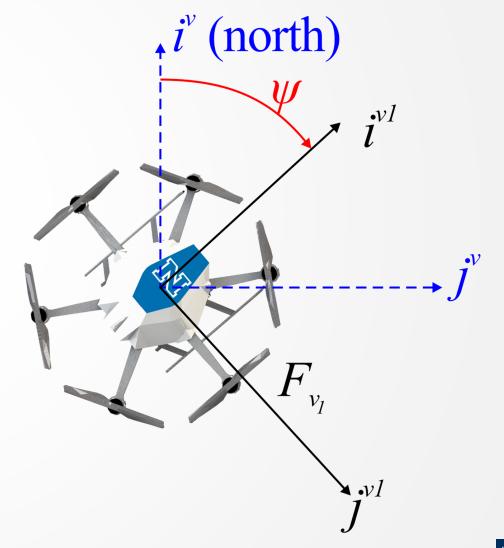
$$\mathcal{R}_b^c \mathcal{R} a^b = \mathcal{R}_a^c$$

### Vehicle-1 Frame

$$\mathbf{p}^{v_1} = \mathcal{R}_v^{v_1} \mathbf{p}^v,$$

$$\mathcal{R}_v^{v_1} = \begin{bmatrix} \cos \psi & \sin \psi & 0 \\ -\sin \psi & \cos \psi & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

 $ightharpoonup \psi$  represents the yaw angle

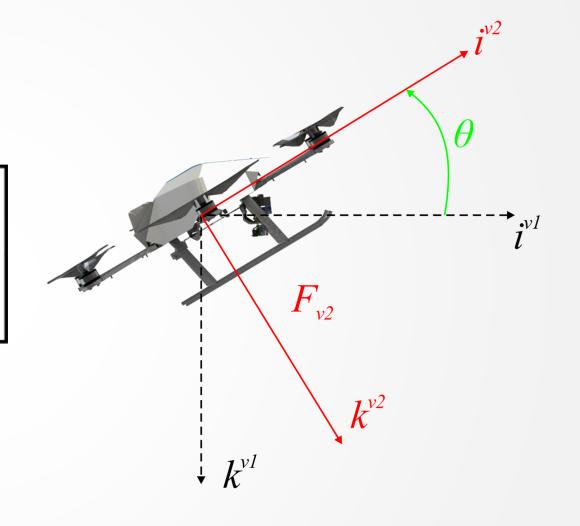


## Vehicle-2 Frame

$$\mathbf{p}^{v_2} = \mathcal{R}_{v_1}^{v_2} \mathbf{p}^{v_1},$$

$$\mathcal{R}_{v_1}^{v_2} = \begin{bmatrix} \cos \theta & 0 & -\sin \theta \\ 0 & 1 & 0 \\ \sin \theta & 0 & \cos \theta \end{bmatrix}$$

ullet heta represents the pitch angle



# **Body Frame**

$$\mathbf{p}^b = \mathcal{R}^b_{v_2} \mathbf{p}^{v_2},$$
 
$$\mathbf{r}^b_{v_2} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos \phi & \sin \phi \\ 0 & -\sin \phi & \cos \phi \end{bmatrix}^{\varphi}$$
 
$$\mathbf{p}^b = \mathbf{r}^b_{v_2} \mathbf{p}^{v_2},$$
 
$$\mathbf{r}^b_{v_2} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos \phi & \sin \phi \\ 0 & -\sin \phi & \cos \phi \end{bmatrix}^{\varphi}$$

# Inertial Frame to Body Frame

Let:

$$\mathcal{R}_{v}^{b}(\phi, \theta, \psi) = \mathcal{R}_{v_{2}}(\phi)\mathcal{R}_{v_{1}}^{v_{2}}(\theta)\mathcal{R}_{v}^{v_{1}}(\psi)$$

$$= \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos\phi & \sin\phi \\ 0 & -\sin\phi & \cos\phi \end{bmatrix} \begin{bmatrix} \cos\theta & 0 & -\sin\theta \\ 0 & 1 & 0 \\ \sin\theta & 0 & \cos\theta \end{bmatrix} \begin{bmatrix} \cos\psi & \sin\psi & 0 \\ -\sin\psi & \cos\psi & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

$$= \begin{bmatrix} c_{\theta}c_{\psi} & c_{\theta}s_{\psi} & -s_{\theta} \\ s_{\phi}s_{\theta}c_{\psi} - c_{\phi}s_{\psi} & s_{\phi}s_{\theta}s_{\psi} + c_{\phi}c_{\psi} & s_{\phi}c_{\theta} \\ c_{\phi}s_{\theta}c_{\psi} + s_{\phi}s_{\psi} & c_{\phi}s_{\theta}s_{\psi} - s_{\phi}c_{\psi} & c_{\phi}c_{\theta} \end{bmatrix}$$

Then:

$$\mathbf{p}^b = \mathcal{R}^b_v \mathbf{p}^v$$

## Relate Translational Velocity-Position

Let [u,v,w] represent the body linear velocities

$$\frac{d}{dt} \begin{bmatrix} p_n \\ p_e \\ p_d \end{bmatrix} = \mathcal{R}_b^v \begin{bmatrix} u \\ v \\ w \end{bmatrix} = (\mathcal{R}_v^b)^T \begin{bmatrix} u \\ v \\ w \end{bmatrix}$$

Which gives:

$$\begin{bmatrix} \dot{p}_n \\ \dot{p}_e \\ \dot{p}_d \end{bmatrix} = \begin{bmatrix} c_{\theta}c_{\psi} & s_{\phi}s_{\theta}c_{\psi} - c_{\phi}s_{\phi} & c_{\phi}s_{\theta}c_{\psi} + s_{\phi}s_{\psi} \\ c_{\theta}s_{\psi} & s_{\phi}s_{\theta}s_{\psi} + c_{\phi}c_{\psi} & c_{\phi}s_{\theta}s_{\psi} - s_{c\psi} \\ -s_{\theta} & s_{\phi}c_{\theta} & c_{c\theta} \end{bmatrix} \begin{bmatrix} u \\ v \\ w \end{bmatrix}$$

# Body Rates – Euler Rates

Let [p,q,r] denote the body angular rates

$$\begin{bmatrix} p \\ q \\ r \end{bmatrix} = \begin{bmatrix} \dot{\phi} \\ 0 \\ 0 \end{bmatrix} + \mathcal{R}_{v_2}^b(\phi) \begin{bmatrix} 0 \\ \dot{\theta} \\ 0 \end{bmatrix} + \mathcal{R}_{v_2}^b(\phi) \mathcal{R}_{v_1}^{v_2}(\theta) \begin{bmatrix} 0 \\ 0 \\ \dot{\psi} \end{bmatrix} = \begin{bmatrix} \dot{\phi} \\ 0 \\ 0 \end{bmatrix} + \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos \phi & \sin \phi \\ 0 & -\sin \phi & \cos \phi \end{bmatrix} \begin{bmatrix} 0 \\ \dot{\theta} \\ 0 \end{bmatrix} + \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos \phi & \sin \phi \\ 0 & -\sin \phi & \cos \phi \end{bmatrix} \begin{bmatrix} \cos \theta & 0 & -\sin \theta \\ 0 & 1 & 0 \\ \sin \theta & 0 & \cos \theta \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ \dot{\psi} \end{bmatrix} = \begin{bmatrix} 1 & 0 & -\sin \theta \\ 0 & \cos \phi & \sin \phi \cos \theta \\ 0 & -\sin \phi & \cos \phi \cos \theta \end{bmatrix} \begin{bmatrix} \dot{\phi} \\ \dot{\theta} \\ \dot{\psi} \end{bmatrix}$$

Inverting this expression:

$$\begin{bmatrix} \dot{\phi} \\ \dot{\theta} \\ \dot{\psi} \end{bmatrix} = \begin{bmatrix} 1 & \sin \phi \tan \theta & \cos \phi \tan \theta \\ 0 & \cos \phi & -\sin \phi \\ 0 & \sin \phi \sec \theta & \cos \phi \sec \theta \end{bmatrix} \begin{bmatrix} p \\ q \\ r \end{bmatrix}$$



# Autonomous Mobile Robot Design Topic: MAV Dynamics

Dr. Kostas Alexis (CSE)

# MAV Dynamics What is the relation between the propeller aerodynamic forces and moments, the gravity force and the motion of the aerial robot?

- Assumption 1: the Micro Aerial Vehicle is flying as a rigid body with negligible aerodynamic effects on it for the employed airspeeds.
- The propeller is considered as a simple propeller disc that generates thrust and a moment around its shaft.



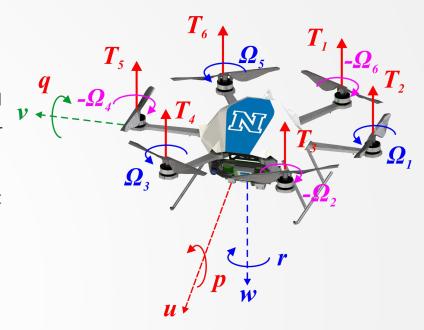
$$F_T = T = C_T \rho A (\Omega R)^2$$
  

$$M_Q = Q = C_Q \rho A (\Omega R)^2 R$$

And let us write:

$$T_i = k_n \Omega_i^2$$

$$M_i = (-1)^{i-1} k_m T_i$$



- Recall the kinematic equations:
  - Translational Kinematic Expression:

$$\begin{bmatrix} \dot{p}_n \\ \dot{p}_e \\ \dot{p}_d \end{bmatrix} = \mathcal{R}_b^v \begin{bmatrix} u \\ v \\ w \end{bmatrix}, \ \mathcal{R}_b^v = \begin{bmatrix} c_\theta c_\psi & s_\phi s_\theta c_\psi - c_\phi s_\psi & c_\phi s_\theta c_\psi + s_\phi s_\psi \\ c_\theta s_\psi & s_\phi s_\theta s_\psi + c_\phi c_\psi & c_\phi s_\theta s_\psi - s_\phi c_\psi \\ -s_\theta & s_\phi c_\theta & c_\phi c_\theta \end{bmatrix}$$

Rotational Kinematic Expression

$$\begin{bmatrix} \dot{\phi} \\ \dot{\theta} \\ \dot{\psi} \end{bmatrix} = \begin{bmatrix} 1 & \sin \phi \tan \theta & \cos \phi \tan \theta \\ 0 & \cos \phi & -\sin \phi \\ 0 & \sin \phi \sec \theta & \cos \phi \sec \theta \end{bmatrix} \begin{bmatrix} p \\ q \\ r \end{bmatrix}$$

To append the forces and moments we need to combine their formulation with

$$\begin{bmatrix} \dot{p}_n \\ \dot{p}_e \\ \dot{p}_d \end{bmatrix} = \mathcal{R}_b^v \begin{bmatrix} u \\ v \\ w \end{bmatrix}, \ \mathcal{R}_b^v = \begin{bmatrix} c_{\theta}c_{\psi} & s_{\phi}s_{\theta}c_{\psi} - c_{\phi}s_{\psi} & c_{\phi}s_{\theta}c_{\psi} + s_{\phi}s_{\psi} \\ c_{\theta}s_{\psi} & s_{\phi}s_{\theta}s_{\psi} + c_{\phi}c_{\psi} & c_{\phi}s_{\theta}s_{\psi} - s_{\phi}c_{\psi} \\ -s_{\theta} & s_{\phi}c_{\theta} & c_{\phi}c_{\theta} \end{bmatrix}$$

$$\begin{bmatrix} \dot{u} \\ \dot{v} \\ \dot{w} \end{bmatrix} = \begin{bmatrix} rv - qw \\ pw - ru \\ qu - pv \end{bmatrix} + \frac{1}{m} \begin{bmatrix} f_x \\ f_y \\ f_z \end{bmatrix}$$

$$\begin{bmatrix} \dot{\phi} \\ \dot{\theta} \\ \dot{\psi} \end{bmatrix} = \begin{bmatrix} 1 & \sin \phi \tan \theta & \cos \phi \tan \theta \\ 0 & \cos \phi & -\sin \phi \\ 0 & \sin \phi \sec \theta & \cos \phi \sec \theta \end{bmatrix} \begin{bmatrix} p \\ q \\ r \end{bmatrix}$$

$$\begin{bmatrix} \dot{p} \\ \dot{q} \\ \dot{r} \end{bmatrix} = \begin{bmatrix} \frac{J_y - J_z}{J_x} qr \\ \frac{J_z - J_x}{J_y} pr \\ \frac{J_x - J_y}{J_z} qr \end{bmatrix} + \begin{bmatrix} \frac{1}{m} M_x \\ \frac{1}{m} M_y \\ \frac{1}{z} M_z \end{bmatrix}$$

Next step: append the MAV forces and moments

MAV forces in the body frame:

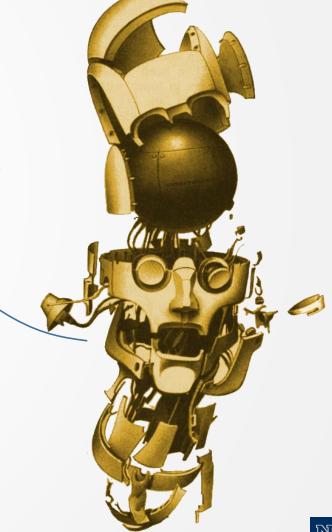
$$\mathbf{f}_b = \begin{bmatrix} f_x \\ f_y \\ f_z \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ \sum_{i=1}^6 T_i \end{bmatrix} - \mathcal{R}_v^b \begin{bmatrix} 0 \\ 0 \\ mg \end{bmatrix}$$

Moments in the body frame:

$$\mathbf{m}_{b} = \begin{bmatrix} M_{x} \\ M_{y} \\ M_{z} \end{bmatrix} = \begin{bmatrix} ls_{30} & l & ls_{30} & -ls_{30} & -l & ls_{30} \\ -lc_{60} & 0 & lc_{60} & lc_{60} & 0 & -lc_{60} \\ -k_{m} & k_{m} & -k_{m} & k_{m} & -k_{m} & k_{m} \end{bmatrix}$$

## Autonomous Robot Challenges

Where am 1? What is my environment?





# Autonomous Mobile Robot Design Topic: Preliminaries on state estimation

Dr. Kostas Alexis (CSE)

# Navigation Sensors

- Providing the capacity to estimate the state of the aerial robot
  - Self-Localize and estimate its pose in the environment
  - Often this requires to also derive the map of the environment
  - In some cases also rely in external systems (e.g. GPS), while a lot of work is undergoing into making aerial robots completely autonomous.





## Classification of Sensors

#### What:

#### Proprioceptive sensors

- Measure values internally to the robot.
  - Angular rate, heading.

#### Exteroceptive sensors

- Information from the robot environment
  - Distances to objects, extraction of features from the environment.



#### How:

#### Passive Sensors

Measure energy coming from a signal of the environment – very much influenced from the environment.

#### Active Sensors

- Emit their proper energy and measure reaction.
- Better performance, but some influence on the environment.
- Not always easily applicable concept.

# Uncertainty Representation

#### Sensing is always related to uncertainties

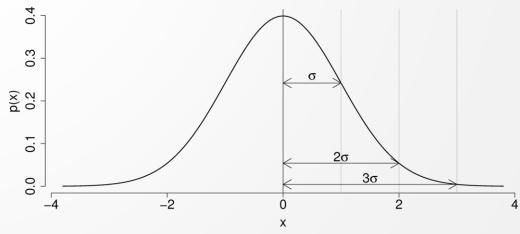
- How can uncertainty be represented or quantified?
- How do they propagate uncertainty of a function of uncertain values?

### Systematic errors

They are caused by factors or processes that can in theory be modeled and, thus, calibrated, (for example the misalignment of a 3-axes accelerometer)

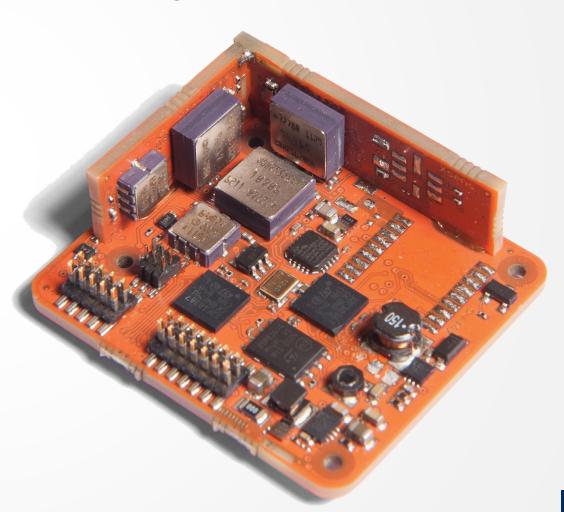
#### Random errors

They cannot be predicted using a sophisticated model but can only be described in probabilistic terms
Normal distribution when μ=0 and σ=1



# Typical Navigation Sensors

- The following sensors are commonly used for the navigation of aerial robots:
  - Inertial Sensors:
    - Accelerometers
    - Gyroscopes
  - Magnetometers (digital compass)
  - Pressure Sensors
    - Barometric pressure for altitude sensing
    - Airspeed measurements
  - GPS
  - Camera based systems
  - Time-of-Flight sensors



# World state (or system state)

- <u>Belief state:</u>
  - Our belief/estimate of the world state
- World state:
  - Real state of the robot in the real world





# Autonomous Mobile Robot Design

Topic: State Estimation – Recap on Probabilities

Dr. Kostas Alexis (CSE)

# Probability theory

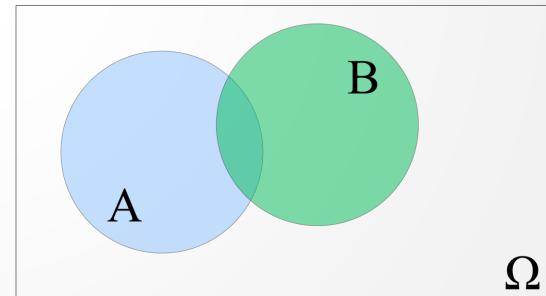
- Random experiment that can produce a number of outcomes, e.g. a rolling dice.
- Sample space, e.g.: {1,2,3,4,5,6}
- Event A is subset of outcomes, e.g. {1,3,5}
- ightharpoonup Probability P(A), e.g. P(A)=0.5

## Axioms of Probability theory

$$-0 \le P(A) \le 1$$

$$P(\Omega) = 1, P(\emptyset) = 0$$

$$P(A \cup B) = P(A) + P(B) - P(A \cap B)$$



### Discrete Random Variables

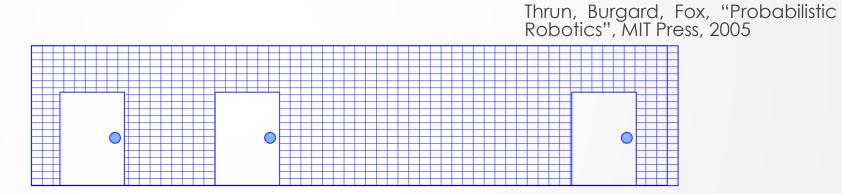
- X denotes a random variable
- $\blacksquare$  X can take on a countable number of values in  $\{x_1, x_2, ..., x_n\}$
- $ightharpoonup P(X=x_i)$  is the probability that the random variable X takes on value  $x_i$
- P(.) is called the probability mass function

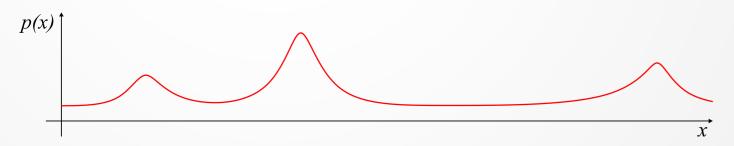
Example: P(Room)=<0.6,0.3,0.06,0.03>, Room one of the office, corridor, lab, kitchen

## Continuous Random Variables

- X takes on continuous values.
- P(X=x) or P(x) is called the **probability density function (PDF)**.

Example:





## Proper Distributions Sum To One

Discrete Case

$$\sum_{x} P(x) = 1$$

Continuous Case 
$$\int p(x) dx = 1$$

## Joint and Conditional Probabilities

• 
$$p(X = x, \text{ and } Y = y) = P(x, y)$$

If X and Y are independent then:

$$P(x,y) = P(x)P(y)$$

Is the probability of x given y

$$P(x|y)P(y) = P(x,y)$$

If X and Y are independent then:

$$P(x|y) = P(x)$$

# Conditional Independence

Definition of conditional independence:

$$P(x,y|z) = P(x|z)P(y|z)$$

Equivalent to:

$$P(x|z) = P(x|y,z)$$

$$P(y|z) = P(y|x,z)$$

Note: this does not necessarily mean that:

$$P(x,y) = P(x)P(y)$$

## Marginalization

Discrete case:

$$P(x) = \sum_{y} P(x, y)$$

Continuous case: 
$$p(x) = \int p(x,y) dy$$

# Marginalization example

P(X,Y)	x1	<b>x</b> 1	<b>x</b> 1	x1	P(Y) ↓
y1	1/8	1/16	1/32	1/32	1/4
y1	1/16	1/8	1/32	1/32	1/4
y1	1/16	1/16	1/16	1/16	1/4
y1	1/4	0	0	0	1/4
P(X) →	1/2	1/4	1/8	1/8	1

## Expected value of a Random Variable

Discrete case: 
$$E[X] = \sum_i x_i P(x_i)$$

Continuous case: 
$$E[X] = \int x P(X=x) dx$$

- The expected value is the weighted average of all values a random variable can take on.
- Expectation is a linear operator:

$$E[aX + b] = aE[X] + b$$

#### Covariance of a Random Variable

Measures the square expected deviation from the mean:

$$Cov[X] = E[X - E[X]]^2 = E[X^2] - E[X]^2$$

### Estimation from Data

 $lackbox{ t Disservations:} \quad \mathbf{x}_1,\mathbf{x}_2,...,\mathbf{x}_n \in \mathcal{R}^d$ 

Sample Mean:  $\mu = \frac{1}{n} \sum_i \mathbf{x}_i$ 

Sample Covariance:

$$\Sigma = \frac{1}{n-1} \sum_{i} (\mathbf{x}_i - \mu)(\mathbf{x}_i - \mu)$$



Autonomous Mobile Robot Design

Topic: State Estimation – Reasoning with Bayes Law

Dr. Kostas Alexis (CSE)

## The State Estimation problem

- We want to estimate the world state x from:
  - Sensor measurements z and
  - Controls u
- ► We need to model the relationship between these random variables, i.e.

$$p(\mathbf{x}|\mathbf{z})$$

$$p(\mathbf{x}'|\mathbf{x},\mathbf{u})$$

## Causal vs. Diagnostic Reasoning

$$P(\mathbf{x}|\mathbf{z})$$
 is diagnostic  $P(\mathbf{z}|\mathbf{x})$  is causal

- Diagnostic reasoning is typically what we need.
- Often causal knowledge is easier to obtain.
- Bayes rule allows us to use causal knowledge in diagnostic reasoning.

## Bayes rule

Definition of conditional probability:

$$P(x,z) = P(x|z)P(z) = P(z|x)P(x)$$

Bayes rule:

Observation likelihood

Prior on world state

$$P(x|z) = \frac{P(z|x)P(x)}{P(z)}$$

Prior on sensor observations

#### Normalization

- lacktriangle Direct computation of P(z) can be difficult.
- Idea: compute improper distribution, normalize afterwards.

- STEP 1: L(x|z) = P(z|x)P(x)
- STEP 2:  $P(z) = \sum_x P(z,x) = \sum_x P(z|x)P(x) = \sum_x L(x|z)$
- STEP 3: P(x|z) = L(x|z)/P(z)

## Example: Sensor Measurement

- Quadrotor seeks the Landing Zone
- The landing zone is marked with many bright lamps
- The quadrotor has a light sensor.



## Example: Sensor Measurement

- Binary sensor  $Z \in \{bright, bright\}$
- ullet Binary world state  $X \in \{home, home\}$
- Sensor model P(Z=bright|X=home)=0.6 P(Z=bright|X=home)=0.3
- lacktriangleright Prior on world state <math>P(X=home)=0.5
- Assume: robot observes light, i.e.  $\,Z=bright\,$
- What is the probability P(X = home | Z = bright) that the robot is above the landing zone.

## Example: Sensor Measurement

Sensor model: P(Z = bright|X = home) = 0.6

$$P(Z = bright|X = \bar{home}) = 0.3$$

- Prior on world state: P(X=home)=0.5
- Probability after observation (using Bayes):

$$P(X = home|Z = bright) = P(bright|home)P(home)$$

$$\frac{P(bright|home)P(home) + P(bright|home)P(home)}{0.6 \cdot 0.5} = 0.67$$



# Autonomous Mobile Robot Design Topic: State Estimation – Bayes Filter

Dr. Kostas Alexis (CSE)

## Markov Assumption

Observations depend only on current state

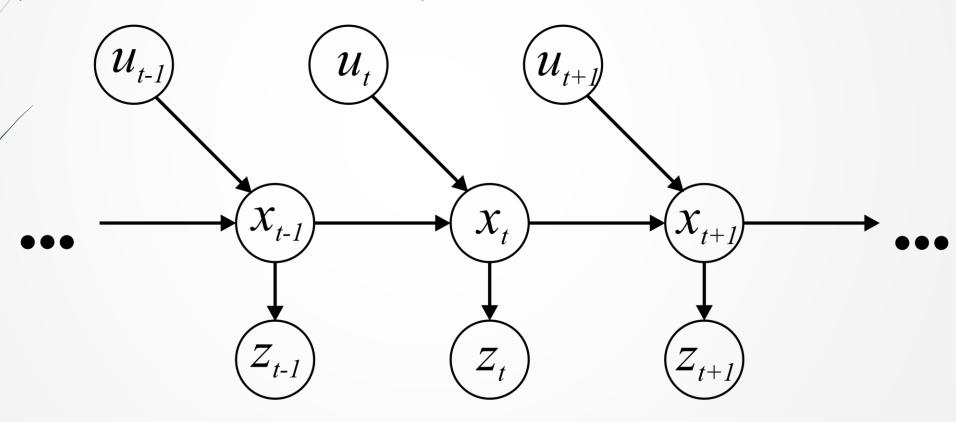
$$P(z_t|x_{0:t}, z_{1:t-1}, u_{1:t}) = P(z_t|x_t)$$

Current state depends only on previous state and current action

$$P(x_t|x_{0:t}, z_{1:t}, u_{1:t}) = P(x_t|x_{t-1}, u_t)$$

## Markov Chain

A Markov Chain is a stochastic process where, given the present state, the past and the future states are independent.



# Underlying Assumptions

- Static world
- Independent noise
- Perfect model, no approximation errors

## Bayes Filter

#### Given

- $begin{array}{c} lacksquare{1} \end{array}$  Sequence of observations and actions:  $z_t, u_t$
- Sensor model: P(z|x)
- ightharpoonup Action model: P(x'|x,u)
- lacktriangleright Prior probability of the system state: P(x)

#### Desired

- lacktriangle Estimate of the state of the dynamic system:  ${oldsymbol{\mathcal{X}}}$
- Posterior of the state is also called belief:

$$Bel(x_t) = P(x_t|u_1, z_1, ..., u_t, z_t)$$

## Bayes Filter Algorithm

- For each time step, do:
  - Apply motion model:

$$\overline{Bel}(x_t) = \sum_{x_t-1} P(x_t|x_{t-1}, u_t)Bel(x_{t-1})$$

Apply sensor model:

$$Bel(x_t) = \eta P(z_t|x_t) \overline{Bel}(x_t)$$

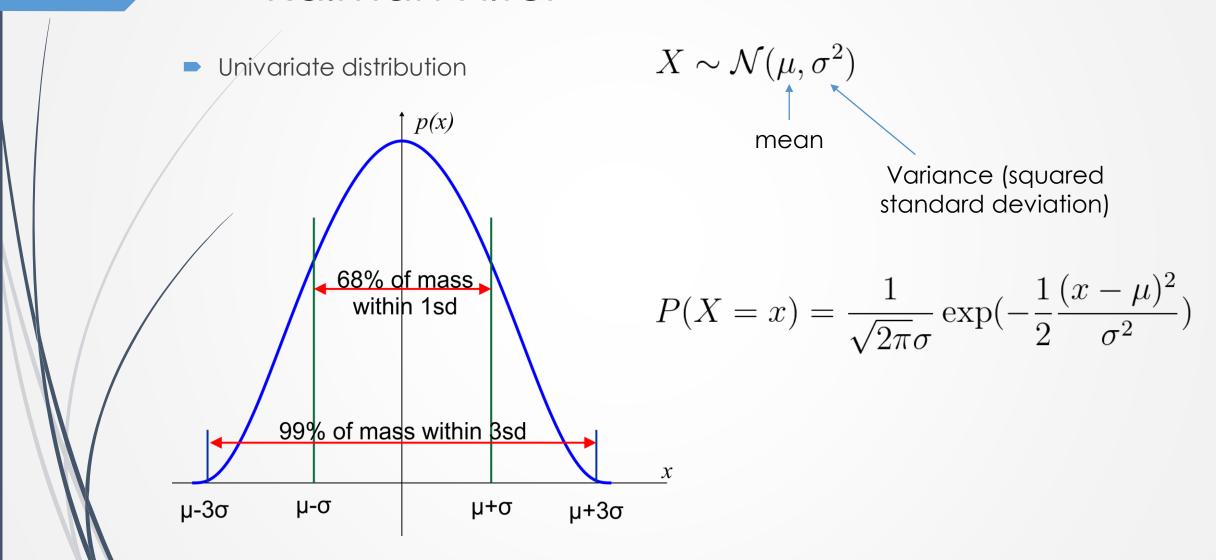
 $\blacksquare$   $\eta$  is a normalization factor to ensure that the probability is maximum 1.



# Autonomous Mobile Robot Design Topic: State Estimation – Kalman Filter

Dr. Kostas Alexis (CSE)

- Bayes filter is a useful tool for state estimation.
- Histogram filter with grid representation is not very efficient.
- How can we represent the state more efficiently?



- Multivariate normal distribution:  $\mathbf{X} \sim \mathcal{N}(\mu, \mathbf{\Sigma})$
- Mean:  $\mu \in \mathcal{R}^n$
- -/Covariance:  $\mathbf{\Sigma} \in \mathbf{R}^{n imes m}$
- Probability density function:

$$p(\mathbf{X} = \mathbf{x}) = \mathcal{N}(\mathbf{x}; \mu, \mathbf{\Sigma}) = \frac{1}{(2\pi)^{n/2} |\mathbf{\Sigma}|^{1/2}} \exp(-\frac{1}{2}(\mathbf{x} - \mu)^T \mathbf{\Sigma}^{-1}(\mathbf{x} - \mu))$$

## Properties of Normal Distributions

Linear transformation – remains Gaussian

$$\mathbf{X} \sim \mathcal{N}(\mu, \mathbf{\Sigma}), \mathbf{Y} \sim \mathbf{A}\mathbf{X} + \mathbf{B}$$
  
 $\Rightarrow \mathbf{Y} \sim \mathcal{N}(\mathbf{A}\mu + \mathbf{B}, \mathbf{A}\mathbf{\Sigma}\mathbf{A}^T)$ 

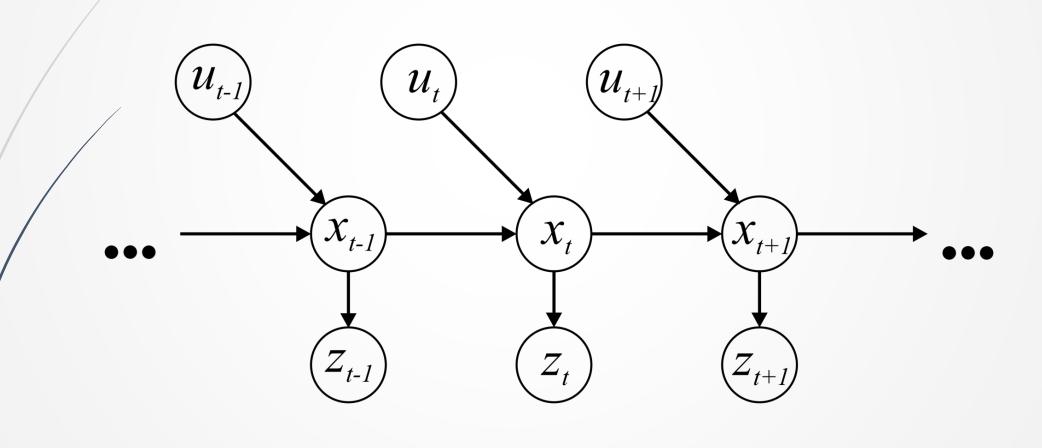
► Intersection of two Gaussians – remains Gaussian

$$\mathbf{X}_1 \sim \mathcal{N}(\boldsymbol{\mu}_1, \boldsymbol{\Sigma}_1), \mathbf{X}_2 \sim \mathcal{N}(\boldsymbol{\mu}_2, \boldsymbol{\Sigma}_2)$$

$$p(\mathbf{X}_1)p(\mathbf{X}_2) = \mathcal{N}\left(\frac{\Sigma_2}{\Sigma_1 + \Sigma_2}\boldsymbol{\mu}_1 + \frac{\Sigma_1}{\Sigma_1 + \Sigma_2}\boldsymbol{\mu}_2, \frac{1}{\Sigma_1^{-1} + \Sigma_2^{-1}}\right)$$

## Linear Process Model

Consider a time-discrete stochastic process (Markov chain)



#### Linear Process Model

- Consider a time-discrete stochastic process
- Represent the estimated state (belief) with a Gaussian

$$\mathbf{x}_t \sim \mathcal{N}(\mu_t, \mathbf{\Sigma}_t)$$

Assume that the system evolves linearly over time, then depends linearly on the controls, and has zero-mean, normally distributed process noise

$$\mathbf{x}_t = \mathbf{A}\mathbf{x}_{t-1} + \mathbf{B}\mathbf{u}_t + \epsilon_t$$

- With  $\epsilon_t \sim \mathcal{N}(\mathbf{0}, \mathbf{Q})$ 

### Linear Observations

 Further, assume we make observations that depend linearly on the state and that are perturbed zero-mean, normally distributed observation noise

$$\mathbf{z}_t = \mathbf{C}\mathbf{x}_t + \delta_t$$

- With  $\delta_t \sim \mathcal{N}(\mathbf{0},\mathbf{R})$ 

Estimates the state  $x_t$  of a discrete-time controlled process that is governed by the linear stochastic difference equation

$$\mathbf{x}_t = \mathbf{A}\mathbf{x}_{t-1} + \mathbf{B}\mathbf{u}_t + \epsilon_t$$

And (linear) measurements of the state

$$\mathbf{z}_t = \mathbf{C}\mathbf{x}_t + \delta_t$$

• With  $\epsilon_t \sim \mathcal{N}(\mathbf{0}, \mathbf{Q})$  and  $\delta_t \sim \mathcal{N}(\mathbf{0}, \mathbf{R})$ 

- lacksquare State  $\mathbf{x} \in \mathbb{R}^n$
- ullet Controls  $\mathbf{u} \in \mathbb{R}^l$
- -/Observations  $\mathbf{z} \in \mathbb{R}^k$
- Process equation  $\mathbf{x}_t = \mathbf{A}\mathbf{x}_{t-1} + \mathbf{B}\mathbf{u}_t + \epsilon_t$
- Measurement equation  $\mathbf{z}_t = \mathbf{C}\mathbf{x}_t + \delta_t$

Initial belief is Gaussian

$$Bel(x_0) = \mathcal{N}(\mathbf{x}_0; \mu_0, \Sigma_0)$$

Next state is also Gaussian (linear transformation)

$$\mathbf{x}_t \sim \mathcal{N}(\mathbf{A}\mathbf{x}_t + \mathbf{B}\mathbf{u}_t, \mathbf{Q})$$

Observations are also Gaussian

$$\mathbf{z}_t \sim \mathcal{N}(\mathbf{C}\mathbf{x}_t, \mathbf{R})$$

## Recall: Bayes Filter Algorithm

- For each step, do:
  - Apply motion model

$$\overline{Bel}(\mathbf{x}_t) = \int p(\mathbf{x}_t | \mathbf{x}_{t-1}, \mathbf{u}_t) Bel(\mathbf{x}_{t-1}) d\mathbf{x}_{t-1}$$

Apply sensor model

$$Bel(\mathbf{x}_t) = \eta p(\mathbf{z}_t | \mathbf{x}_t) \overline{Bel}(\mathbf{x}_t)$$

## From Bayes Filter to Kalman Filter

- For each step, do:
  - Apply motion model

$$\overline{Bel}(\mathbf{x}_t) = \int \underbrace{p(\mathbf{x}_t | \mathbf{x}_{t-1}, \mathbf{u}_t)}_{\mathcal{N}(\mathbf{x}_t; \mathbf{A}\mathbf{x}_{t-1} + \mathbf{B}\mathbf{u}_k t, \mathbf{Q})} \underbrace{Bel(\mathbf{x}_{t-1})}_{\mathcal{N}(\mathbf{x}_{t-1}; \mu_{t-1}, \mathbf{\Sigma}_{t-1})} d\mathbf{x}_{t-1}$$

$$= \mathcal{N}(\mathbf{x}_t; \mathbf{A}\mu_{t-1} + \mathbf{B}\mathbf{u}_t, \mathbf{A}\mathbf{\Sigma}\mathbf{A}^T + \mathbf{Q})$$

$$= \mathcal{N}(\mathbf{x}_t; \bar{\mu}_t, \bar{\Sigma}_t)$$

## From Bayes Filter to Kalman Filter

- For each step, do:
  - Apply sensor model

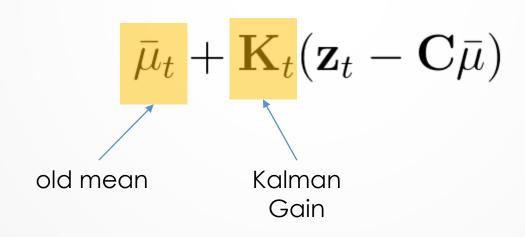
$$\overline{Bel}(\mathbf{x}_t) = \eta \underbrace{p(\mathbf{z}_t | \mathbf{x}_t)}_{\mathcal{N}(\mathbf{z}_t; \mathbf{C}\mathbf{x}_t, \mathbf{R})} \underline{Bel}(\mathbf{x}_t) 
\mathcal{N}(\mathbf{z}_t; \mathbf{C}\mathbf{x}_t, \mathbf{R}) \mathcal{N}(\mathbf{x}_t; \bar{\mu}_t, \bar{\Sigma}_t) 
= \mathcal{N}(\mathbf{x}_t; \bar{\mu}_t + \mathbf{K}_t(\mathbf{z}_t - \mathbf{C}\bar{\mu}), (\mathbf{I} - \mathbf{K}_t)\mathbf{C})\bar{\Sigma}) 
= \mathcal{N}(x_t; \mu_t, \Sigma_t)$$

• With 
$$\mathbf{K}_t = ar{\mathbf{\Sigma}}_t \mathbf{C}^T (\mathbf{C} ar{\mathbf{\Sigma}}_t \mathbf{C}^T + \mathbf{R})^{-1}$$
 (Kalman Gain)

## From Bayes Filter to Kalman Filter

Blends between our previous estimate  $\bar{\mu}_t$  and the discrepancy between our sensor observations and our predictions.

The degree to which we believe in our sensor observations is the Kalman Gain. And this depends on a formula based on the errors of sensing etc. In fact it depends on the ratio between our uncertainty  $\Sigma$  and the uncertainty of our sensor observations R.



## Kalman Filter Algorithm

Prediction & Correction steps can happen in any order.

- For each step, do:
  - Apply motion model (prediction step)

$$ar{m{\mu}}_t = \mathbf{A}m{\mu}_{t-1} + \mathbf{B}\mathbf{u}_t \ ar{m{\Sigma}}_t = \mathbf{A}m{\Sigma}\mathbf{A}^{ op} + \mathbf{Q}$$

Apply sensor model (correction step)

$$oldsymbol{\mu}_t = ar{oldsymbol{\mu}}_t + \mathbf{K}_t(\mathbf{z}_t - \mathbf{C}ar{oldsymbol{\mu}}_t) \ oldsymbol{\Sigma}_t = (\mathbf{I} - \mathbf{K}_t\mathbf{C})ar{oldsymbol{\Sigma}}_t$$

• With 
$$\mathbf{K}_t = ar{\mathbf{\Sigma}}_t \mathbf{C}^ op (\mathbf{C} ar{\mathbf{\Sigma}}_t \mathbf{C}^ op + \mathbf{R})^{-1}$$

## Kalman Filter Algorithm

Prediction & Correction steps can happen in any order.

#### **Prediction**

$$ar{oldsymbol{\mu}}_t = \mathbf{A}oldsymbol{\mu}_{t-1} + \mathbf{B}\mathbf{u}_t \ ar{oldsymbol{\Sigma}}_t = \mathbf{A}oldsymbol{\Sigma}\mathbf{A}^ op + \mathbf{Q}$$

#### Correction

$$egin{aligned} oldsymbol{\mu}_t &= ar{oldsymbol{\mu}}_t + \mathbf{K}_t (\mathbf{z}_t - \mathbf{C} ar{oldsymbol{\mu}}_t) \ oldsymbol{\Sigma}_t &= (\mathbf{I} - \mathbf{K}_t \mathbf{C}) ar{oldsymbol{\Sigma}}_t \ \mathbf{K}_t &= ar{oldsymbol{\Sigma}}_t \mathbf{C}^{ op} (\mathbf{C} ar{oldsymbol{\Sigma}}_t \mathbf{C}^{ op} + \mathbf{R})^{-1} \end{aligned}$$



# Autonomous Mobile Robot Design Topic: Extended Kalman Filter

Dr. Kostas Alexis (CSE)

These slides relied on the lectures from C. Stachniss, J. Sturm and the book "Probabilistic Robotics" from Thurn et al.

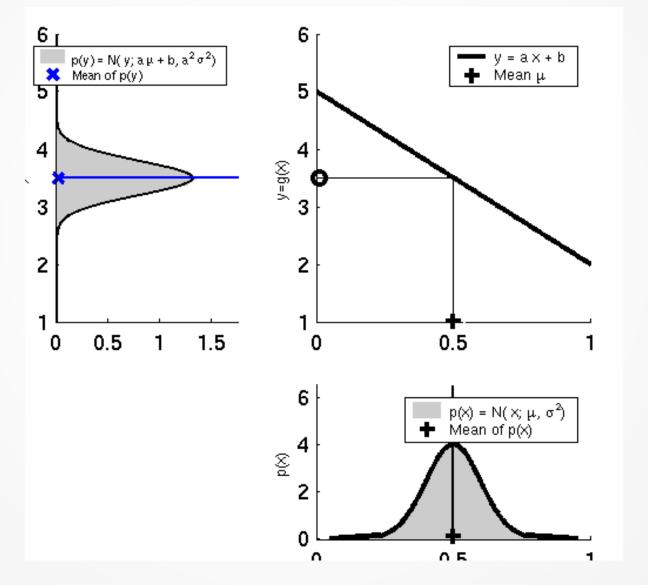
# Kalman Filter Assumptions

- Gaussian distributions and noise
- Linear motion and observation model
  - What if this is not the case?

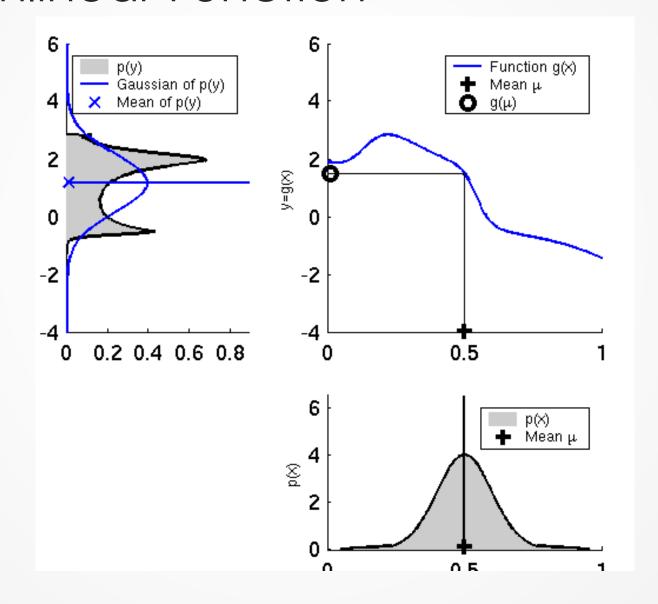
$$x_t = A_t x_{t-1} + B_t u_t + \epsilon_t$$

$$z_t = C_t x_t + \delta_t$$

# Linearity Assumption Revisited



# Nonlinear Function



# Nonlinear Dynamical Systems

- Real-life robots are mostly nonlinear systems.
- The motion equations are expressed as nonlinear differential (or difference) equations:

$$x_t = g(u_t, x_{t-1})$$

Also leading to a nonlinear observation function:

$$z_t = h(x_t)$$

# Taylor Expansion

- Solution: approximate via linearization of both functions
- Motion Function:

$$g(x_{t-1}, u_t) \approx g(\mu_{t-1}, u_t) + \frac{\partial g(\mu_{t-1}, u_t)}{\partial x_{t-1}} (x_{t-1} - \mu_{t-1})$$
$$= g(\mu_{t-1}, u_t) + G_t(x_{t-1} - \mu_{t-1})$$

Observation Function:

$$h(x_t) \approx h(\bar{\mu}_t) + \frac{\partial h(\bar{\mu}_t)}{\partial x_t} (x_t - \mu_t)$$
$$= h(\bar{\mu}_t) + H_t(x_t - \mu_t)$$

#### Reminder: Jacobian Matrix

- It is a non-square matrix mxn in general
- ► Given a vector-valued function:

$$g(x) = \begin{pmatrix} g_1(x) \\ g_2(x) \\ \vdots \\ g_m(x) \end{pmatrix}$$

The **Jacobian matrix** is defined as:

$$G_{x} = \begin{pmatrix} \frac{\partial g_{1}}{\partial x_{1}} & \frac{\partial g_{1}}{\partial x_{2}} & \dots & \frac{\partial g_{1}}{\partial x_{n}} \\ \frac{\partial g_{2}}{\partial x_{1}} & \frac{\partial g_{2}}{\partial x_{2}} & \dots & \frac{\partial g_{2}}{\partial x_{n}} \\ \vdots & \vdots & \ddots & \vdots \\ \frac{\partial g_{m}}{\partial x_{1}} & \frac{\partial g_{m}}{\partial x_{2}} & \dots & \frac{\partial g_{m}}{\partial x_{n}} \end{pmatrix}$$

#### Extended Kalman Filter

- For each time step, do:
- Apply Motion Model:

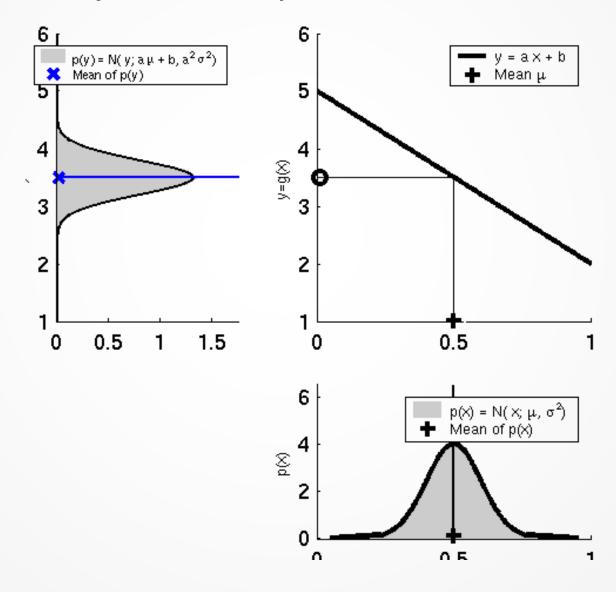
$$ar{\mu}_t = g(\mu_{t-1}, u_t)$$
 $ar{\Sigma}_t = G_t \Sigma G_t^\top + Q$  with  $G_t = \frac{\partial g(\mu_{t-1}, u_t)}{\partial x_{t-1}}$ 

Apply Sensor Model:

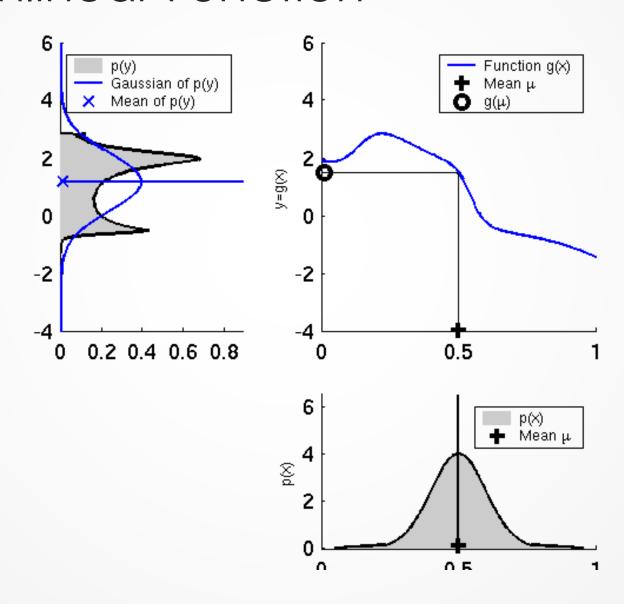
$$\mu_t = \bar{\mu}_t + K_t(z_t - h(\bar{\mu}_t))$$
  
$$\Sigma_t = (I - K_t H_t) \bar{\Sigma}_t$$

where 
$$K_t = \bar{\Sigma}_t H_t^{\top} (H_t \bar{\Sigma}_t H_t^{\top} + R)^{-1}$$
 and  $H_t = \frac{\partial h(\bar{\mu}_t)}{\partial x_t}$ 

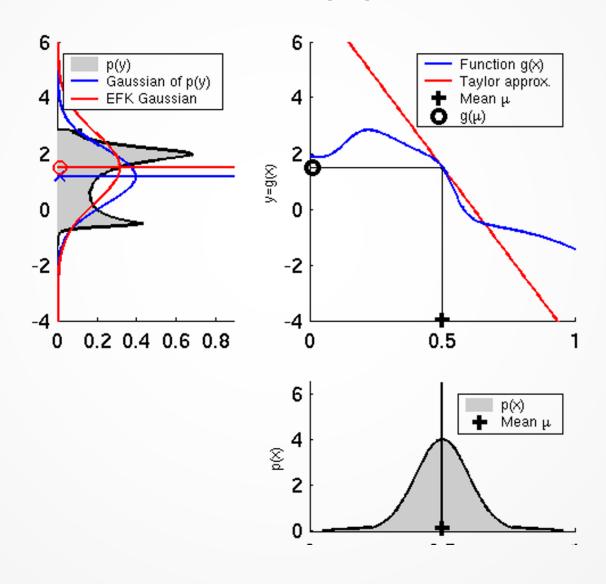
# Linearity Assumption Revisited



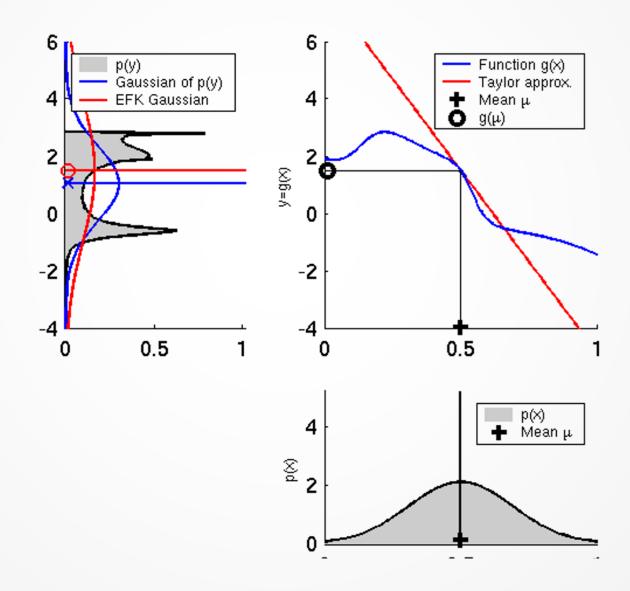
# Nonlinear Function



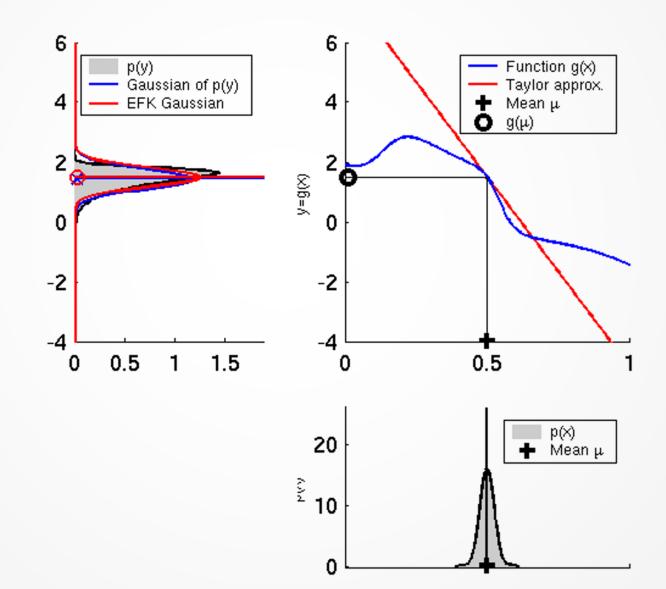
# EKF Linearization (1)



# EKF Linearization (2)



# EKF Linearization (3)



#### Linearized Motion Model

The linearized model leads to:

$$p(x_t \mid u_t, x_{t-1}) \approx \det(2\pi R_t)^{-\frac{1}{2}}$$

$$\exp\left(-\frac{1}{2} (x_t - g(u_t, \mu_{t-1}) - G_t (x_{t-1} - \mu_{t-1}))^T\right)$$

$$R_t^{-1} (x_t - g(u_t, \mu_{t-1}) - G_t (x_{t-1} - \mu_{t-1}))$$
linearized model

 $lacktriangleright R_t$  describes the noise of the motion.

## EKF Algorithm

1: Extended\_Kalman\_filter(
$$\mu_{t-1}, \Sigma_{t-1}, u_t, z_t$$
):

2: 
$$\bar{\mu}_t = g(u_t, \mu_{t-1})$$

2: 
$$\bar{\mu}_t = \underline{g}(u_t, \mu_{t-1})$$
  
3:  $\bar{\Sigma}_t = G_t \; \Sigma_{t-1} \; G_t^T + R_t$ 

$$A_t \leftrightarrow G_t$$

4: 
$$K_t = \bar{\Sigma}_t H_t^T (H_t \bar{\Sigma}_t H_t^T + Q_t)^{-1}$$
  $C_t \leftrightarrow H_t$ 

5: 
$$\mu_t = \bar{\mu}_t + K_t(z_t - \underline{h}(\bar{\mu}_t))$$

6: 
$$\Sigma_t = (I - K_t H_t) \Sigma_t$$

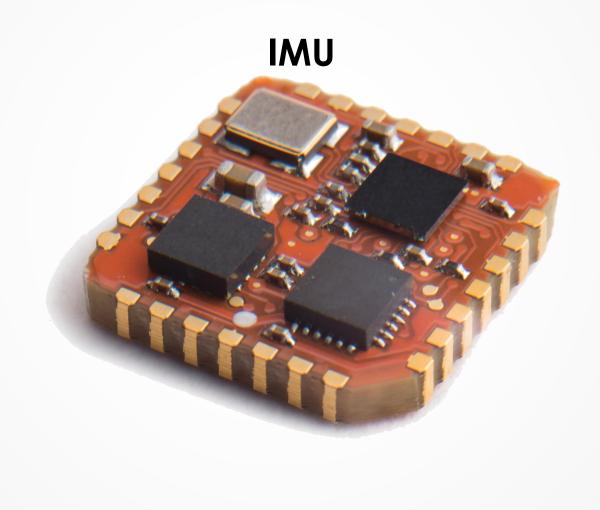
7: return 
$$\mu_t, \Sigma_t$$

KF vs EKF

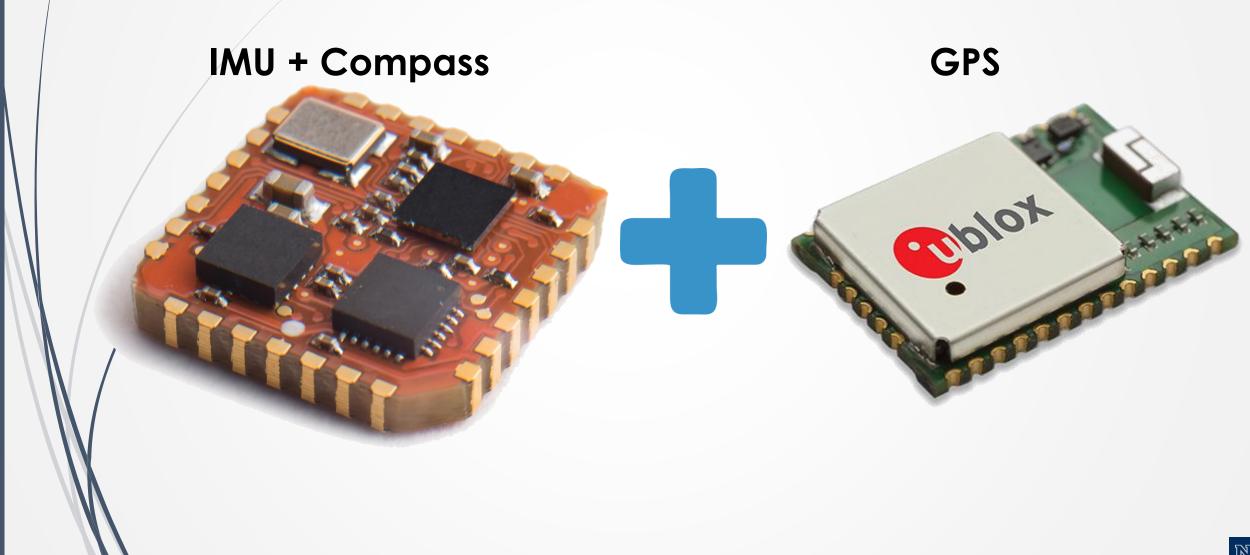
# **EKF Summary**

- Extension of the Kalman Filter.
- One way to deal with nonlinearities.
- Performs local linearizations.
- Works well in practice for moderate nonlinearities.
- Large uncertainty leads to increased approximation error.

# **EKF** Discussion

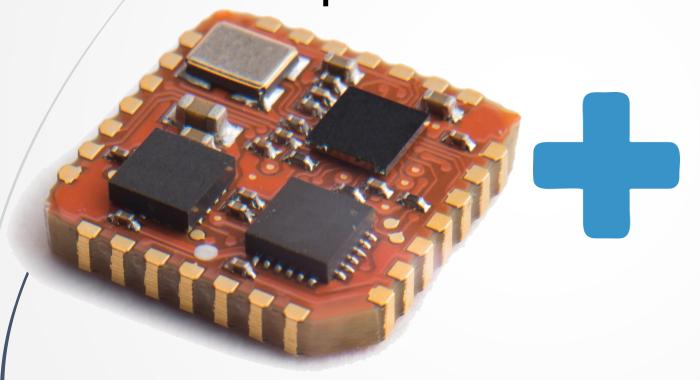


## **EKF** Discussion



## **EKF** Discussion





#### Camera



